

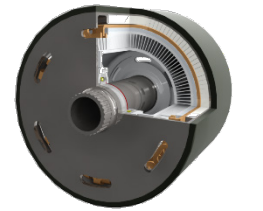
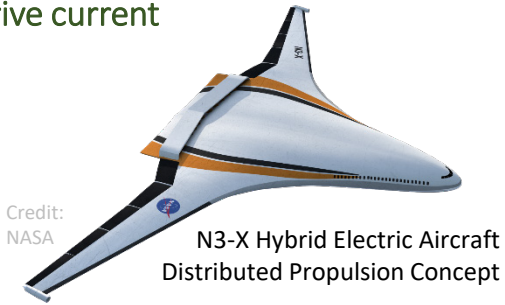
# Extreme Performance Scalable Inverter Architecture for More Electric Aircraft (MEA) Propulsion



Berkeley Power and Energy Center

## Motivation and Application

- Commercial aviation benefits from electric & hybrid vehicles
- Electric engines can be quieter and cleaner than jet engines
- Electric drive system must be **power-dense** and **efficient**
- Advanced power dense motors<sup>2</sup> need **low THD, high frequency drive current**



University of Illinois  
1 MW Motor Concept<sup>2</sup>

### NASA MEA Roadmap<sup>1</sup>

**NOx emissions**

80%

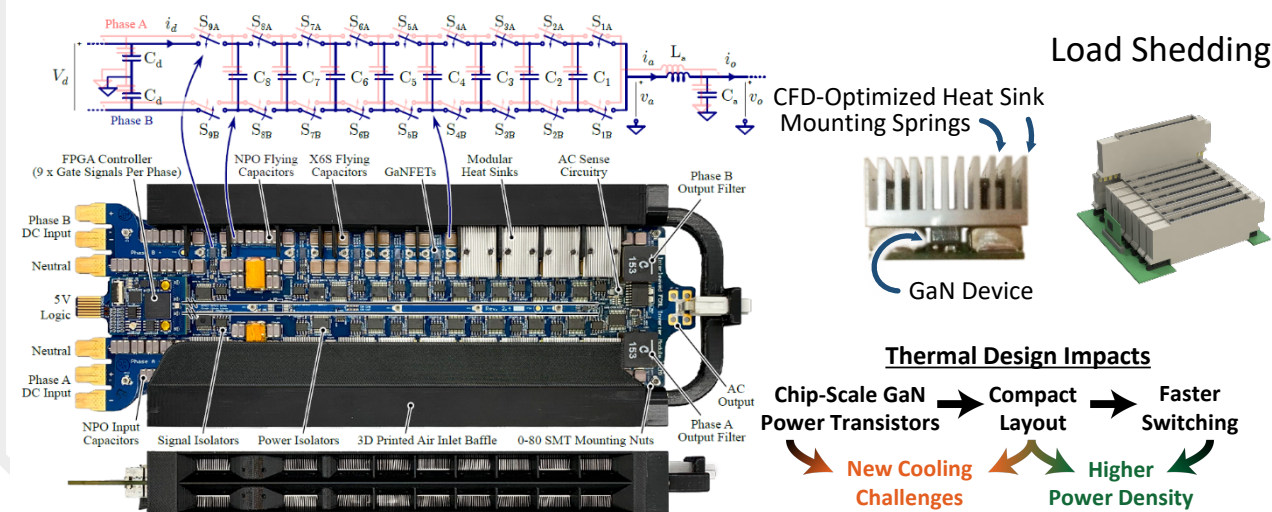
**fuel consumption**

60%

**acoustic noise**

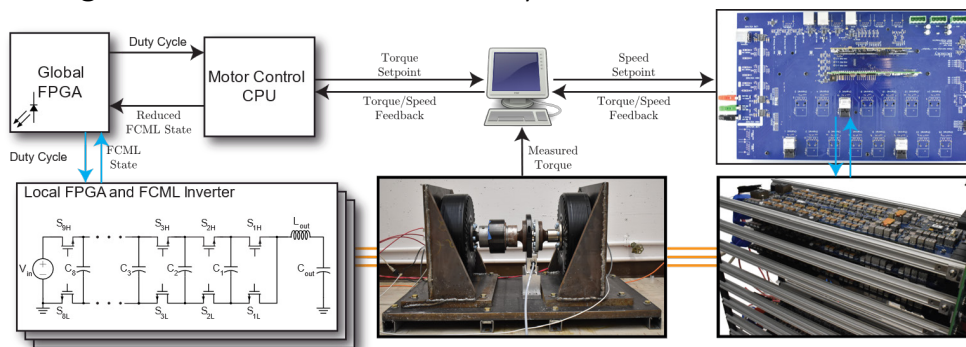
71dB

## Electrical, Mechanical and Thermal Management Hardware



## High Power Dynamometer

- Dyno incorporates two low-inductance Emrax 348 machines (peak power: 260 kW)
- Testing validated the Flying Capacitor Multilevel Converter's (FCML) strength in a realistic motor drive system



## Experimental Verification or Other

- Prototype meets NASA performance metrics for turbo-electric aircraft<sup>4</sup>
- Integration of advanced thermal management will boost maximum output power and efficiency
- Modular design provides for power scalability and fault resiliency
- Next steps: verification of floating-point motor control algorithm and high-power dyno and next generation inverter hardware development

	NASA Target	This Work
<b>Peak Efficiency</b>	<b>99%</b>	<b>98.95%</b>
<b>Power Density</b>	<b>19 kW/kg</b>	<b>38.4 kW/kg</b>

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